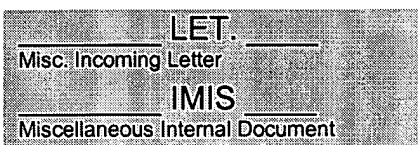




A DOCPHOENIX

**INCOMING**

\_\_\_\_\_ ACPA \_\_\_\_\_  
 Continuing Prosecution Application  
 \_\_\_\_\_ AP.B \_\_\_\_\_  
 Appeal Brief  
 \_\_\_\_\_ C680 \_\_\_\_\_  
 Request for Corrected Notice/Allowance  
 \_\_\_\_\_ C.AD \_\_\_\_\_  
 Change of Address  
 \_\_\_\_\_ CFILE \_\_\_\_\_  
 Request for Corrected Filing Receipt  
 \_\_\_\_\_ COCIN \_\_\_\_\_  
 Papers filed re Certificate of Corrections  
 \_\_\_\_\_ CRFD \_\_\_\_\_  
 Computer Readable Form Defective  
 \_\_\_\_\_ CRFE \_\_\_\_\_  
 Computer Readable Form 'ENTERED'  
 \_\_\_\_\_ EABN \_\_\_\_\_  
 Request for Express Abandonment  
 \_\_\_\_\_ ELC. \_\_\_\_\_  
 Response to Election/Restriction  
 \_\_\_\_\_ IFEE \_\_\_\_\_  
 Issue Fee Transmittal PTOL 85 B  
 \_\_\_\_\_ IRFND \_\_\_\_\_  
 Refund Request  
 \_\_\_\_\_ L\_RIN \_\_\_\_\_  
 Any Incoming to L&R  
 \_\_\_\_\_ N417 \_\_\_\_\_  
 Copy of EFS Receipt Acknowledgement  
 \_\_\_\_\_ N/AP \_\_\_\_\_  
 Notice of Appeal  
 \_\_\_\_\_ PA.. \_\_\_\_\_  
 Change in Power of Attorney  
 \_\_\_\_\_ PC/I \_\_\_\_\_  
 Power to Make Copies or to Inspect  
 \_\_\_\_\_ PEF. \_\_\_\_\_  
 Pre-Exam Formalities Response  
 \_\_\_\_\_ PEFRRREISS \_\_\_\_\_  
 Pre-Exam Formalities Reissue Response  
 \_\_\_\_\_ PEFRSEQ \_\_\_\_\_  
 Pre-Exam Formalities Sequence Reply

**INCOMING**

\_\_\_\_\_ PGEA \_\_\_\_\_  
 Req Express Aband to avoid Publication  
 \_\_\_\_\_ PGA9 \_\_\_\_\_  
 Req for Corrected Pat App Publication  
 \_\_\_\_\_ PGREF \_\_\_\_\_  
 Req for Refund of Publication Fee Paid  
 \_\_\_\_\_ PROTEST \_\_\_\_\_  
 Protest Documents Filed by 3<sup>rd</sup> Party  
 \_\_\_\_\_ PROTRANS \_\_\_\_\_  
 Translation of Provisional in Nonprovisional  
 \_\_\_\_\_ REM \_\_\_\_\_  
 Applicant Remarks in Amendment  
 \_\_\_\_\_ RESC \_\_\_\_\_  
 Rescind Non-Publication Request  
 \_\_\_\_\_ ROCKET \_\_\_\_\_  
 Request for Design Processing  
 \_\_\_\_\_ XT/ \_\_\_\_\_  
 Extension of Time filed separate

**APPL PARTS**

\_\_\_\_\_ 371P \_\_\_\_\_  
 PCT Papers in a 371 Application  
 \_\_\_\_\_ A... \_\_\_\_\_  
 Amendment Including Elections  
 \_\_\_\_\_ A.NE \_\_\_\_\_  
 After Final Amendment  
 \_\_\_\_\_ A.PE \_\_\_\_\_  
 Preliminary Amendment  
 \_\_\_\_\_ ABST \_\_\_\_\_  
 Abstract  
 \_\_\_\_\_ ADS \_\_\_\_\_  
 Application Data Sheet  
 \_\_\_\_\_ AF/D \_\_\_\_\_  
 Affidavit or Exhibit Received  
 \_\_\_\_\_ APPENDIX \_\_\_\_\_  
 Appendix

**APPL PARTS**

\_\_\_\_\_ ARTIFACT \_\_\_\_\_  
 Artifact  
 \_\_\_\_\_ CLM \_\_\_\_\_  
 Claim  
 \_\_\_\_\_ COMPUTER \_\_\_\_\_  
 Computer Program Listing  
 \_\_\_\_\_ CRFL \_\_\_\_\_  
 CRF Transfer Request  
 \_\_\_\_\_ CRFS \_\_\_\_\_  
 Computer Readable Form Statement  
 \_\_\_\_\_ DIST \_\_\_\_\_  
 Terminal Disclaimer Filed  
 \_\_\_\_\_ DRW \_\_\_\_\_  
 Drawings  
 \_\_\_\_\_ FOR \_\_\_\_\_  
 Foreign Reference  
 \_\_\_\_\_ FRPR \_\_\_\_\_  
 Foreign Priority Papers  
 \_\_\_\_\_ IDS \_\_\_\_\_  
 IDS Including 1449  
 \_\_\_\_\_ NPL \_\_\_\_\_  
 Non-Patent Literature  
 \_\_\_\_\_ OATH \_\_\_\_\_  
 Oath or Declaration  
 \_\_\_\_\_ PET. \_\_\_\_\_  
 Petition  
 \_\_\_\_\_ RUSH \_\_\_\_\_  
 OPUBS Printer Query  
 \_\_\_\_\_ SEQLIST \_\_\_\_\_  
 Sequence Listing  
 \_\_\_\_\_ SPEC \_\_\_\_\_  
 Specification  
 \_\_\_\_\_ SPEC NO \_\_\_\_\_  
 Specification Not in English

IN THE CLAIMS

Please amend the claims as follows:

Claim 1 (Currently Amended) A turbine moving blade comprising a platform having a gas path surface extending in the combustion gas flow direction, and a blade portion erecting on said platform, said gas path surface of platform being coated with a thermal barrier coating, wherein

said thermal barrier coating substantially covering said gas path surface and being is formed so as to go around from said gas path surface of platform to at least a part of the outer peripheral face of said platform.

Claim 2 (Original) The turbine moving blade according to claim 1, wherein a step portion is formed in at least a part of the peripheral edge portion of said platform, and said thermal barrier coating is formed so that it goes around to said step portion and the end face thereof is in contact with the upper face of said step portion.

Claim 3 (Currently Amended) A turbine moving blade comprising a platform, a blade portion erecting on said platform, and a shroud provided at the tip end of said blade portion, a gas path surface extending in the combustion gas flow direction of said shroud being coated with a thermal barrier coating, wherein

said thermal barrier coating substantially covering said gas path surface and being is formed so as to go around from said gas path surface of shroud to at least a part of the outer peripheral face of said shroud.

Claim 4 (Original) The turbine moving blade according to claim 3, wherein a step portion is formed in at least a part of the peripheral edge portion of said shroud, and said

thermal barrier coating is formed so that it goes around to said step portion and the end face thereof is in contact with the upper face of said step portion.

Claim 5 (Currently Amended) A turbine stationary blade comprising a pair of shrouds each having a gas path surface extending in the combustion gas flow direction, and a blade portion held between said shrouds, at least either one of said shrouds being coated with a thermal barrier coating, wherein

said thermal barrier coating substantially covering said gas path surface and being is formed so as to go around from said gas path surface of shroud to at least a part of the outer peripheral face of said shroud.

Claim 6 (Original) The turbine stationary blade according to claim 5, wherein a step portion is formed in at least a part of the peripheral edge portion of said shroud, and said thermal barrier coating is formed so that it goes around to said step portion and the end face thereof is in contact with the upper face of said step portion.

Claim 7 (Currently Amended) A turbine split ring having a gas path surface extending in the combustion gas flow direction, said gas path surface being coated with a thermal barrier coating, wherein

said thermal barrier coating substantially covering said gas path surface and being is formed so as to go around from said gas path surface to at least a part of the outer peripheral face.

Claim 8 (Original) The turbine split ring according to claim 7, wherein a step portion is formed in at least a part of the peripheral edge portion, and said thermal barrier coating is

formed so that it goes around to said step portion and the end face thereof is in contact with the upper face of said step portion.

Claim 9 (Currently Amended) A gas turbine for producing power by expanding a high-temperature and high-pressure combustion gas by using a turbine stationary blade and a turbine moving blade, wherein

said turbine moving blade comprises a platform having a gas path surface extending in the combustion gas flow direction, a blade portion erecting on said platform, and a thermal barrier coating for covering said gas path surface of platform, and said thermal barrier coating substantially covering said gas path surface and being is formed so as to go around from said gas path surface to at least a part of the outer peripheral face of said platform.

Claim 10 (Currently Amended) A gas turbine for producing power by expanding a high-temperature and high-pressure combustion gas by using a turbine stationary blade and a turbine moving blade, wherein

said turbine moving blade comprises a platform, a blade portion erecting on said platform, a shroud provided at the tip end of said blade portion, and a thermal barrier coating for covering a gas path surface extending in the combustion gas flow direction of said shroud, and said thermal barrier coating substantially covering said gas path surface and being is formed so as to go around from said gas path surface of shroud to at least a part of the outer peripheral face of said shroud.

Claim 11 (Currently Amended) A gas turbine for producing power by expanding a high-temperature and high-pressure combustion gas by using a turbine stationary blade and a turbine moving blade, wherein